# Equation of state of shock-compressed liquids: Carbon dioxide and air 

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#### Abstract

Equation-of-state data were measured for liquid carbon dioxide and air shock-compressed to pressures in the range $28-71 \mathrm{GPa}$ ( $280-710 \mathrm{kbar}$ ) using a two-stage light-gas gun. The experimental methods are described. The data indicate that shock-compressed liquid $\mathrm{CO}_{2}$ decomposes at pressures above 34 GPa . Liquid air dissociates above a comparable shock pressure, as does liquid nitrogen. Theoretical intermolecular potentials are derived for $\mathrm{CO}_{2}$ from the data. The calculated shock temperature for the onset of $\mathrm{CO}_{2}$ decomposition is 4500 K at a volume of $17 \mathrm{~cm}^{3} / \mathrm{mol}$.


## I. INTRODUCTION

Shock compression of molecular liquids produces high densities and temperatures which induce a variety of phenomena. For example, carbonaceous molecules, such as CO, decompose and react to form mixtures of chemical species. ${ }^{1}$ Diatomic molecules, such as $\mathrm{N}_{2}$ and $\mathrm{O}_{2}$, dissociate. ${ }^{2,3}$ The properties of $\mathrm{CO}_{2}, \mathrm{~N}_{2}$, and $\mathrm{O}_{2}$ and their mixtures are of interest because of their presence in reacted explosives. In addition species containing $\mathrm{C}, \mathrm{O}$, and N are present at high densities and temperature in the interiors of the giant planets Uranus and Neptune. Experimental equation-of-state data are required to develop reliable theoretical descriptions of molecular systems at high pressures and temperatures. Such descriptions are useful for developing models of planetary interiors and reacted explosives, for example.

Carbon dioxide has been single-shocked to $63 \mathrm{GPa}(630$ kbar) using solid specimens. ${ }^{4}$ Liquid specimens were singleshocked to 29 GPa and double-shocked to $56 \mathrm{GPa} .^{5}$ Solid specimens have been compressed statically as high as 51 $\mathrm{GPa} .^{6-8}$ All these data are in the range of stability of the $\mathrm{CO}_{2}$ molecule. In this paper we report equation-of-state results for liquid $\mathrm{CO}_{2}$ single-shocked in the range $28-71 \mathrm{GPa}$. The purpose was to look for the onset of molecular decomposition at higher pressures and temperatures than achieved previously. Equimolar mixtures of liquid $\mathrm{N}_{2}$ and $\mathrm{O}_{2}$ were sin-gle-shocked to 24 GPa and double-shocked to $66 \mathrm{GPa} .{ }^{9}$ We report herein results for liquid air single-shocked in the range $28-69 \mathrm{GPa}$, which is within the range of dissociation of both $\mathrm{N}_{2}$ and $\mathrm{O}_{2} .{ }^{2,3,10-12}$ The data reported here indicate that shocked liquid $\mathrm{CO}_{2}$ begins to decompose above 34 GPa and liquid air dissociates above a comparable shock pressure.

## II. EXPERIMENT

Equation-of-state data were measured using a method based on the Rankine-Hugoniot equations, which conserve mass, momentum, and energy across the front of a shock wave. The equations are

$$
\begin{equation*}
P-P_{0}=\rho_{0}\left(u_{s}-u_{0}\right)\left(u_{p}-u_{0}\right) \tag{1}
\end{equation*}
$$

[^0]\[

$$
\begin{align*}
& V=V_{0}\left[1-\left(u_{p}-u_{0}\right) /\left(u_{s}-u_{0}\right)\right]  \tag{2}\\
& E-E_{0}=0.5\left(P+P_{0}\right)\left(V-V_{0}\right) \tag{3}
\end{align*}
$$
\]

where $P_{0}, \rho_{0}, E_{0}$, and $u_{0}$ are, respectively, the initial pressure, density, specific internal energy, and mass velocity of the material ahead of the shock front; $\mathrm{V}_{0}=1 / \rho_{0}$ is the initial specific volume; $P, V$, and $E$ are, respectively, the final shock pressure, specific volume, and specific internal energy; $u_{s}$ is the velocity of the shock wave; and $u_{p}$ is the mass velocity of the material behind the shock front.

The experiments involve measurements of the velocity of impact $u_{I}$ of a metal plate and of the velocity of the shock wave $u_{s}$ induced in a specimen. ${ }^{13-15}$ The plate, of Ta or A1 alloy 1100 in these experiments, is accelerated to several $\mathrm{km} / \mathrm{s}$ by a two-stage light-gas gun. Impact velocity is measured by flash x-radiography and shock velocity is measured with electrical shock-arrival-time detectors.

Liquid specimens were prepared by condensing highpurity gaseous $\mathrm{CO}_{2}$ or dry air ( $78.1 \% \mathrm{~N}_{2}, 21.0 \% \mathrm{O}_{2}$, and $0.9 \% \mathrm{Ar}$ ) into a precooled specimen holder. The specimen holder is positioned on the axis of the gun in the evacuated target chamber. For the $\mathrm{CO}_{2}$ experiments the initial specimen conditions were chosen to be essentially the same as those used by Schott. ${ }^{5}$ The Al specimen holder was controlled at a temperature of $\sim 218 \mathrm{~K}$ by control of the flow of cold $\mathrm{N}_{2}$ gas. ${ }^{16}$ Initial specimen temperature was measured with a calibrated Pt resistance thermometer. The condensed specimen was pressurized to $\sim 7$ bar with $\mathrm{CO}_{2}$ gas to achieve the liquid phase. Initial specimen pressure was measured with a capacitance manometer. To obtain minimal distortion of the impact wall or base plate of the liquid specimen holder, caused by the 7 bar differential pressure across it, a 3 mm thick Al alloy 1100 base plate was used. The electrical shock detectors were positioned on radii of 5.0 and 8.5 mm with a detector on axis, as previously. ${ }^{14}$ The separation of the two planes on which the detectors are positioned was 2.0 mm . Metal impactor plates were 25 mm in diameter. For the liquid-air experiments liquid $\mathrm{N}_{2}$ was used as the coolant. ${ }^{14}$ Initial specimen temperature was taken to be the temperature on the nitrogen saturation curve at the measured barometric pressure. The Al alloy 1100 base plate was 2 mm thick. For $\mathrm{CO}_{2}$ and air initial specimen pressures were set $\sim 900$ and 50 Torr, respectively, above saturation pressure to suppress boiling and bubbles in the specimens. Initial

TABLE I. Coefficients $C$ and $S$, where $u_{s}=C+S u_{p}$, for the shock-wave equations of state of Al and Ta , including coefficients of Al corrected for the 77 K initial temperature of the liquid air experiments.

| Metal | $T_{0}(\mathrm{~K})$ | $C(\mathrm{~cm} / \mu \mathrm{s})$ | $S$ |
| :---: | :---: | :---: | :---: |
| Al | 300 | 0.5386 | 1.339 |
| Al | 77 | 0.5420 | 1.334 |
| Ta | 300 | 0.3291 | 1.308 |

specimen density was taken to be that of the published saturation curve at initial specimen temperature for $\mathrm{CO}_{2},{ }^{17}$ and air. ${ }^{18}$

Shock pressure, density, and specific internal energy are derived from the measured impactor and shock velocities and initial specimen density by the shock-impedance-matching process and the Rankine-Hugoniot Eqs. (1)-(3). The shock-wave equations of state of the Al (Ref. 19) and Ta (Refs. 19 and 20) impactor plates and the cold-Al (Ref. 14) base plate of the specimen holder, which were used in the shock-impedance-matching process, ${ }^{14,15}$ are listed in Table I. These shock-wave equations of state are represented by a linear relation between shock and mass velocity: $u_{s}=C+S u_{p}$.

## III. EXPERIMENTAL RESULTS

The single-shock equation-of-state data of liquid $\mathrm{CO}_{2}$ and air are listed in Table II. Shock velocity vs mass velocity are plotted for $\mathrm{CO}_{2}$ in Fig. 1. The $\mathrm{CO}_{2}$ data of Schott ${ }^{5}$ are also plotted in Fig. 1. The two data sets agree in the overlap region near shock velocity of $7 \mathrm{~km} / \mathrm{s}$. Our data point at 28 GPa is in good agreement with the linear fit to the $u_{s}-u_{p}$ data of Schott in the $u_{p}$ range $2.0-3.6 \mathrm{~km} / \mathrm{s}: u_{s}=C+S u_{p}$, where $C=1.69 \mathrm{~km} / \mathrm{s}$ and $S=1.40$. The data for $\mathrm{CO}_{2}$ plotted in Fig. 1 show a change in slope at $u_{p}=4 \mathrm{~km} / \mathrm{s}$, which corresponds to a shock pressure of 34 GPa . Such a slope change indicates the onset of chemical decomposition, as observed in $\mathrm{CS}_{2},{ }^{21} \mathrm{CO},{ }^{1}$ and $\mathrm{N}_{2},{ }^{2,15}$ for example.

Shock velocity vs mass velocity are plotted for liquid air in Fig. 2. For comparison the corresponding fit to the data of liquid nitrogen, ${ }^{15}$ which composes $78 \%$ of air, is also plotted in Fig. 2. The $u_{s}-u_{p}$ data of air in Fig. 2 are quite similar to


FIG. 1. Shock velacity vs mass velocity for liquid carbon dioxide. Circles are this work; squares are from Ref. 5. Dashed line is extrapolation of linear fit to data of Ref. 5.
those of nitrogen and suggest that the nitrogen molecules in air dissociate above about 30 GPa , as in pure nitrogen. ${ }^{14}$ Dissociation occurs at a similar shock pressure in liquid oxygen. ${ }^{3}$

## IV. THEORETICAL CALCULATIONS FOR $\mathrm{CO}_{2}$

As described in Sec. III, the experimental data for $\mathrm{CO}_{2}$ indicate shock-induced dissociation at pressures above $\sim 34$ GPa. This phenomena can be characterized with a theoretical calculation using the Gibb's free energy $G\left(P, T,\left\{n_{i}\right\}\right)$, which includes both $\mathrm{CO}_{2}$ and its dissociation products. The calculations presented here utilize five gaseous species ( $\mathrm{CO}_{2}, \mathrm{CO}, \mathrm{O}_{2}, \mathrm{O}, \mathrm{O}_{3}$ ) and three condensed forms of carbon (liquid, graphite, and diamond). The condensed phases are

TABLE II. Hugoniot data for liquid $\mathrm{CO}_{2}$ and air. The initial densities of the Al (alloy 1100 ) impactors were 2.708 and $2.716 \mathrm{~g} / \mathrm{cm}^{3}$ for the $\mathrm{CO}_{2}$ and air experiments, respectively. The initial density of the Al (alloy 1100 ) base plates were 2.718 and $2.747 \mathrm{~g} / \mathrm{cm}^{3}$ for the $\mathrm{CO}_{2}$ and air experiments, respectively, with thermal contraction taken into account. The initial density of the Ta impactors was $16.66 \mathrm{~g} / \mathrm{cmi}^{7}$. The average initial molar volumes were 37.55 and 32.76 $\mathrm{cm}^{3} / \mathrm{mol}$ for $\mathrm{CO}_{2}$ and air, respectively.

| Shot | Impactor | $\begin{gathered} u_{I} \\ (\mathrm{~km} / \mathrm{s}) \end{gathered}$ | $\begin{gathered} T_{0} \\ (\mathrm{~K}) \end{gathered}$ | $\begin{gathered} \rho_{0} \\ \left(\mathrm{~g} / \mathrm{cm}^{3}\right) \end{gathered}$ | $\begin{gathered} u_{s} \\ (\mathrm{~km} / \mathrm{s}) \end{gathered}$ | $\begin{gathered} u_{p} \\ (\mathrm{~km} / \mathrm{s}) \end{gathered}$ | $\begin{gathered} P \\ (\mathrm{GPa})^{a} \end{gathered}$ | $\begin{gathered} V \\ \left(\mathrm{~cm}^{3} / \mathrm{mol}\right) \end{gathered}$ |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| CO 23 | Al | 4.928 | 218.8 | 1.171 | $6.609 \pm 0.040$ | $3.556 \pm 0.021$ | $27.52 \pm 0.24$ | $17.36 \pm 0.18$ |
| CO22A | Al | 6.451 | 218.4 | 1.172 | $7.912 \pm 0.033$ | $4.549 \pm 0.026$ | $42.19 \pm 0.32$ | $15.96 \pm 0.16$ |
| CO21 | Ta | 5.143 | 218.5 | 1.172 | $8.645 \pm 0.049$ | $5.382 \pm 0.038$ | $54.53 \pm 0.53$ | $14.17 \pm 0.23$ |
| CO 24 | Ta | 6.070 | 218.5 | 1.172 | $9.646 \pm 0.053$ | $6.264 \pm 0.049$ | $70.82 \pm 0.74$ | $13.16 \pm 0.25$ |
| AIR5 | Al | 5.706 | 77.1 | 0.884 | $7.442+0.052$ | $4.317+0.025$ | $28.40 \pm 0.28$ | $13.76 \pm 0.18$ |
| AIR2 | Ta | 5.236 | 77.1 | 0.884 | $9.200 \pm 0.138$ | $5.788 \pm 0.050$ | $47.07 \pm 0.80$ | $12.15 \pm 0.35$ |
| AIR4 | Ta | 6.752 | 77.1 | 0.884 | $10.56 \pm 0.08$ | $7.379 \pm 0.062$ | $68.86 \pm 0.85$ | $9.86 \pm 0.28$ |

[^1]

FIG. 2. Shock velocity vs mass velocity for liquid-air data (circles) and fit to liquid-nitrogen data (solid curve from Ref. 15).
included to allow some dissociated carbon atoms to precipitate under shock pressure. We minimize $G\left(P, T,\left\{n_{i}\right\}\right)$ with respect to the compositions $\left\{n_{i}\right\}$ of the various species at fixed pressure $P$ and temperature $T$. The resulting $G\left(P, T,\left\{n_{i}\right\}\right)$ is used to compute thermodynamic quantities subject to the constraint of the Rankine-Hugoniot relations, Eqs. (1)-(3).

The minimization of the Gibb's free energy was carried out by the chemical equilibrium (CHEQ) code. ${ }^{22}$ Briefly, for


FIG. 3. Comparison of theoretical Hugoniots of $\mathrm{CO}_{2}$ with experimental data of this work and with data of Ref. 5. Parameter sets are explained in text.
the condensed phases the code uses our three-phase model ${ }^{23}$ of carbon which takes into account all relevant experimental and theoretical information on graphite, diamond, and liquid carbon. For gaseous mixtures it uses soft-sphere perturbation theory ${ }^{24}$ and an improved mixture model ${ }^{25}$ based on the van der Waals one-fluid theory. Our calculations employ the exponential-six (exp-6) potential with three parameters $\epsilon, r^{*}$, and $\alpha$. Since $\mathrm{CO}_{2}$ is the dominant species in the present case, the discussion below will be concerned with the question of finding optimum parameter sets for $\mathrm{CO}_{2}$. The CO parameters used in our calculations are described in Ref. 1 and those for $\mathrm{O}_{2}$ and O are a slight modification of the parameters used in our earlier analysis of the shock data of liquid $\mathrm{O}_{2} .{ }^{3}$ The parameters for $\mathrm{O}_{3}$ are estimated from the corresponding-states scaling relation with $\alpha=13 .^{26} \mathrm{~A}$ detailed theoretical account on the variation of the exp-6 parameters will be described in a separate paper. ${ }^{27}$

Our discussions are based on calculations with four sets of exp-6 parameters for $\mathrm{CO}_{2}$ :

$$
\begin{array}{lll}
\text { Set } \mathrm{A}: ~ & \epsilon / k=245.6 \mathrm{~K}, & r^{*}=4.200 \AA, \\
\text { Set } \mathrm{B}: ~ & \epsilon / k=335.0 \mathrm{~K}, & r^{*}=4.096 \AA, \\
\text { Set } \mathrm{C}: ~ & \epsilon / k=335.0 \mathrm{~K}, & r^{*}=4.096 \AA, \\
\text { Set } \mathrm{D}: & \epsilon / k=245.6 \mathrm{~K}, & r^{*}=4.170 \AA, \\
\AA & \alpha=13.78, & \alpha=13.5
\end{array}
$$

where $k$ is the Boltzmann constant. Sets A and D are our present modifications of an earlier parameter set ${ }^{27}$ $\left(\epsilon / k, r^{*}, \alpha\right)=(245.6 \mathrm{~K}, 4.17 \AA, 13)$, which was obtained by scaling the exp-6 parameters of argon. ${ }^{28}$ Set $D$ has a larger $\alpha$ (13.5) than the earlier set. Set A is made still stiffer by increasing $r^{*}$. Set C is that of Shaw and Johnson, ${ }^{29}$ who modified their earlier set, ${ }^{30}\left(\epsilon / k, r^{*}, \alpha\right)=(338.3 \mathrm{~K}, 4.0935 \AA$, 14.40025 ), to fit the data of Schott. ${ }^{5}$

Figure 3 presents the $P-V$ Hugoniots calculated from sets A, B, and C. They are compared with the experimental data in Sec. III and those of Schott. Set B has a lower $\alpha$ (13.5) than the value (13.78) used for set C. This was done to give a somewhat improved fit to the experimental data between 30 and 40 GPa . However, all these sets can be considered reasonable to 40 GPa . One important outcome from these calculations is that, at pressures above 40 GPa , a significant proportion of carbon atoms in $\mathrm{CO}_{2}$ molecules will dissociate and condense into the liquid phase and that it gives rise to a shoulder-like structure seen in the experimental data near 40 GPa . If the $\mathrm{CO}_{2}$ molecules are prohibited from the shock dissociation, the resulting Hugoniot will be much stiffer than the experimental data. Figure 3 includes such an example obtained from set A. Figure 4 shows the shock temperatures predicted from the parameter sets used in Fig. 3. As expected, the dissociation of $\mathrm{CO}_{2}$ lowers considerably the shock temperature. Note that, in spite of sizable differences in values of $\epsilon$ and $r^{*}$, sets $\mathbf{A}$ and $\mathbf{B}$ both predict nearly identical shock pressure and temperature. This implies that, in selecting an optimum $\mathrm{CO}_{2}$ potential to represent the conditions of interest here, other experimental data, such as double-shock data, are needed in addition to the singleshock data considered in this work.

It should be emphasized that the exp-6 potentials are effective two-body potentials, which spherically average the


FIG. 4. Shock temperatures vs. shock pressures calculated with parameter sets used in Fig. 3.
angular dependence of molecular interactions. This approximation has been found to be adequate ${ }^{30}$ for Hugoniot calculations at conditions such as those found in our experiments, where the temperatures are very high and comparable to or exceed the barriers to rotation.

In previous work which examined the shock data of simple molecules, such as Ar (Ref. 31) and $\mathrm{He},{ }^{32,33}$ we found that the intermolecular potentials determined by fitting to the shock data and static high-pressure solid data were generally in good agreement. In Fig. 5 we made a similar test for $\mathrm{CO}_{2}$. Set D described earlier fits to the 299.15 K solid iso-


FIG. 5. Static isotherms of $\mathrm{CO}_{2}$ at 299 K calculated using various exp-6 parameter sets compared with experimental data of Ref. 7.
therm data of Liu. ${ }^{7}$ However, the same set produces a Hugoniot which is somewhat softer than the experimental shock data. On the other hand, set B which fits the experimental Hugoniot data predicts pressures higher than the solid isotherm above 15 GPa . Set A behaves similar to set B to $\sim 30 \mathrm{GPa}$, beyond which it predicts slightly lower pressures. However, both sets $A$ and $B$ give pressures much higher than the experimental isotherm above 15 GPa . Set D , not shown, predicts even higher pressures. This is not surprising since $\mathrm{CO}_{2}$ is a long linear molecule and the $\mathrm{CO}_{2}$ molecules in the solid state at low temperatures and high pressures orient themselves in a geometry which minimizes their intermolecular repulsions. In comparison, high temperatures (Fig. 4) created by shock compression allow the $\mathrm{CO}_{2}$ molecules to access some repulsive orientations, which are otherwise difficult under the low-temperature static environment of Liu's experiment. Thus, an effective spherical interaction (such as set A), which averages all orientations, will have too large a repulsion for low energy configurations, which in turn predicts pressures higher than the experimental solid isotherm.

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[^1]:    ${ }^{2} 1 \mathrm{GPa}=10 \mathrm{kbar}$.

